Hints to selected problems

(c) Chris Ormel; see lecture notes part II for the exercises

Hand in HW-problems by Mar. 15 13:00 to Jacob Arcangeli

Exercise 1.13

Use conservation of angular momentum and energy. Shown in class.

Exercise 1.14

This is a straightforward but rather math-heavy problem. The goal is to find the precise deflection angle of a gravitational scattering (hyperbolic encounter).

- (a) Note that derivatives (primes) are with respect to θ , not to time!
- (b) Somewhat mathy, but straightforward.
- (c) The boundary conditions that you should apply are those at the beginning where $v_r = -v_{\infty}$ and $\theta = \pi$.
- (d) **Please do (e) first**. This is a rather algebraically-heavy exercise that you perhaps want to do on a rainy Sunday afternoon. Substitute $\sin \theta = \sqrt{1 \cos \theta^2}$, isolate the square-root, and solve the quadratic equation. The answer is:

$$\theta = \pm \arccos \frac{b_{90}}{b_{90} + R} \tag{1}$$

As important than deriving this expression is to inspect that the limiting cases $(b_{90} \ll R)$ and $b_{90} \gg R$ make sense.

(e) Look at the solution for a hint.

Exercise 1.15 (HW) [4pt]

For this exercise, order-of-magnitude expressions suffice! Discard numerical constant and use ~ instead of equal signs, *ie*. $M \sim \rho_{\bullet} R^3$, $\pi \sim 1$, etc. Note that question (c) has been split in two and that we ask you to produce some plots.

- (a) Here you should derive the given de/dt expression due to visous stirring. Consider the following:
 - the dispersion-dominated regime holds: the approach velocities (here Δv , in class also referred to as v_{∞}) are determined by the eccentric motion of the planetesimal.
 - inclinations are similar to eccentricities, $i \simeq e/2$.

Substitute $GM_{\star} = a^3 \Omega^2$ at some point.

- (b) Back to aerodynamics. Because we are considering planetesimals, you can safely assume that C_D is order unity and that the drag force is quadratic in velocity: $F_{\text{drag}} = -\frac{1}{2}C_DA\rho_{\text{gas}}(\Delta v)^2$ with Δv now the relative motion with respect to the gas. You can assume that Δv is again due to the eccentricic motion of the planetesimal and ignore the sub-Keplerian motion of the gas; in other words, you can assume that $ev_K \gg \eta v_K$ (which is not always the case).
- (c) Constant means here that Θ is independent of the mass (or radius) of the embryo, assuming constant internal density. The Safronov factor will depend on the location in the disk and on \mathcal{F}_{aero} (properties of the planetesimals). Also, the question should read **10 Earth-mass** planets (not size!).
- (d) In this question you are asked to make some figures. First, choose your favorite "realistic" disk surface density profile, Σ_{gas}(r), Σ_{solids}(r), T(r) where r is the disk orbital radius. From the disk profile you can further calculate ρ_{gas} and h_{gas} at any point r. You can assume that planetesimals represent the bulk of the solid mass (rather than the embryos). Finally, choose a planetesimal size s_P somewhere between 1 and 1,000 km. Other constants (internal densities, b̃) are up to you. Just mention the parameters that you have adopted.

Now that we have an expression for the Safronov number Θ from (c), you can calculate the growth timescale t_{growth} , which is a function of embryo mass, disk orbital radius and the planetesimal (aerodynamical) properties. This is of course the growth timescale without focusing divided by the Safronov factor:

$$t_{\text{growth}} \sim \frac{R\rho_{\bullet}}{\Sigma\Omega} \frac{1}{\Theta}$$
 (2)

With your favorite plotting tool, plot as function of disk orbital radius:

- t_{growth} as function of disk orbital radius r for an embryo mass of 10 Earth masses.
- The embryo mass M_p that corresponds to a growth time of 10 Myr.

Please return clear plots with logarithmic axes, large-enough fonts and label them correctly!

Exercise 1.16 (HW) [2pt]

Note that in this exercise "PA" means pebble accretion, while "Saf" refers to the previously-derived planetesimal accretion.

- (a)
- (b)
- (c) You should give an expression for R/R_{Hill} in which either *R* (the radius of the embryo) or *M* (its mass) does *not* appear. Fill in appropriate numbers and give the dependence of R/R_{Hill} as function of orbital distance *r*.
- (d) For planetesimals, the zero-eccentricity limit always implies the shear-dominated regime. This maximizes the focusing factor. Express Θ_{Saf} in terms of R_{Hill}/R . Then compare.

Exercise 1.17 (HW-Bonus) [1pt]

Here we have again assumed the shear-dominated regime for pebble accretion, where the impact parameter b_{col} is large. We also assumed that the pebbles have settled to the midplane, which renders the system 2D. Hence, the form of \dot{M} .

Exercise 1.18

(a) You should of course use that $P = c_s^2 \rho$ where c_s is constant.

(b)

(c) We can just divide the above expression by $4\pi\rho_c R_c^3/3$:

$$\frac{M_{\text{atm}}}{M_c} = \frac{3R_c\rho_{\text{neb}}}{R_b\rho_c} \exp\left[\frac{R_b}{R_c}\right] = \frac{3\rho_{\text{neb}}}{\rho_c} \frac{\exp[x]}{x}$$
(3)

where $x = R_b/R_c$. Of course $\rho_{\text{neb}}/\rho_c \ll 1$ but this gets compensated by already moderate x.

Exercise 1.19

In this adiabatic case case c_s and the temperature are no longer constant. Therefore, R_b is defined in terms of the nebular parameters: $R_b = GM_p/c_{neb}^2$.

Exercise 1.20 (HW) [3pt]

(a)

- (b) The *M* that appears in T(r) and $\rho(r)$ is the total mass, M_{tot} .
- (c)
- (d) We already discussed this partly in class. Also give a physical explanation for the dependence of the critical core mass on the parameter *W*.

Exercise 1.21 (HW-Bonus) [2pt]

Unfortunately, we won't be able to discuss orbit crossing in our Friday lecture. This exercise is therefore bonus. I have also re-arranged this exercise. Take care. You can assume $\Delta = 10$ as was discussed in the text. Furthermore take an internal density of 5 g cm⁻³ for the embryos. Assume that resonances are not applicable; planets positions are not correlated. Take a purely probabilistic approach. Naturally, an order-of-magnitude approach suffices.

- (a) In the answer *r* denotes the "radius", but it would in fact be more precise to replace it with the semi-major axis *a*. The key to the solution is to calculate the fraction of the orbit where a collision becomes possible. **Also:** convert the derived probabilities into a collision time t_{coll} for both cases. Express your answers symbolically in terms of the orbital period t_K , R_{12} *i* and *a*.
- (b) Plug in numbers for both the 2D and 3D cases. For the inclinations you can assume $i \sim e$, although this is incorrect (see below).
- (c) Accounting for gravitational focusing, how would this change the 2D **and** 3D timescales, derived above?
- (d) Additional question: Why is the assumption $i \sim e$ invalid for this case? Remark that initially (after the gas of the disk has dissapated) the embryos start out in well-separated, non-crossing orbits and can be found in approximately the same orbital plane.

Exercise 2.1

(a) **HW-BONUS.** [1pt] Errata: the numerical constant in front of equation (2.3) $[\frac{2}{3}]$ is correct! In contrast to what was previously mentioned. Note that Δa cannot be negative; hence $\Delta a = |a_1 - a_0|$. Hint: do a Taylor-approximation on $\Omega(a)$ around the semi-major axis of one of the bodies (say, the one of $a = a_0$):

$$\Omega_K(a) \approx \Omega_K(a_0) + \left(\frac{d\Omega}{da}\right)_{a_0} (a - a_0) \tag{4}$$

This will tell you the relative angular speed at which the two bodies separate. (The rest should really be straightforward...)

(b) There should be no *a* in eq. (2.4b). Because $e \ll 1$, $M \approx v$ (but not quite).